

Technical Information
 This map was compiled from data acquired during an airborne electromagnetic survey carried out by CGO using a HELITEM™ Time Domain Electromagnetic (TDEM) system. The system was mounted on a Eurocopter AS350B3 helicopter (registration C-PKAX) and was carried out between April 17th and April 20th, 2015. The aircraft flight elevation was maintained at a nominal ground clearance of 12 m. Aircraft navigation used a 12 channel Real Time Kinematic (RTK) GPS. The flight differential correction was subsequently applied to finalize flight path position. A vertically mounted video camera was used to record images of the ground. The radar height was recorded for times per second using a Sironix CS 2 osum-vapor magnetometer. The magnetic attitude was recorded ten times per second using a Minotaur attitude transducer. The magnetic data were recorded 10 times per second using a Sironix CS 2 osum-vapor magnetometer.

Survey Area Parameters

Traverse	45°/225°
Traverse line azimuth	250 m
Traverse line spacing	1300'±10'
Tie line spacing	1000 m
Aircraft nominal clearance	83 m
EM transmitter nominal clearance	35 m
Magnetic sensor nominal clearance	35 m
EM Receiver nominal clearance	83 m

Electromagnetics
 The TDEM system operating at a base frequency of 30 Hz transmits a 4 ms time-varying signal from a two turn, 108 m horizontal loop mounted approximately 47.0 m above and 12.5 m behind the aircraft. The configuration provides a dipole moment of 1.84 x 10⁷ Am². The response of conductors in the subsurface is sampled 2048 times per second. The system uses a 4 m x 2 m x 2 m electrically shielded cabin and a 20 m x 20 m x 2 m electrically shielded set of reference magnetic data. The EM system records data in a continuous stream for each of the three components. A 3-second response stack was used before outputting the data as 30 times windows at a final recording rate of 10 times per second. The EM receiver records the change in the magnetic field with respect to the 108 m from which the secondary total magnetic field (B) is numerically integrated. High-altitude background sections from the start and end of each flight allow a 1st-order removal of system drift.

Electromagnetic System Specifications

Base Frequency	30 Hz
Waveform	Half sine wave
Pulse width	4 ms
Transmitter Area	795 m ² (2 turns)
Transmitter Off-time	125 ms
Transmitter Loop	30 m diameter
Transmitter Current	1300 A
(Loop moment approximately)	1.84 x 10 ⁷ Am ² (±1°C)
Windowed data sampling rate	10 Hz
Receiver	3-component induction coil (X, Y, Z)
Voltage (dBV)	
Receiver Response	
Digital recording	All raw data channels (30 channels)
1 st off-time channel	Channel 3 at ~1.160 ms after pulse turn on
Tie-line Configuration	Towed transmitter below receiver

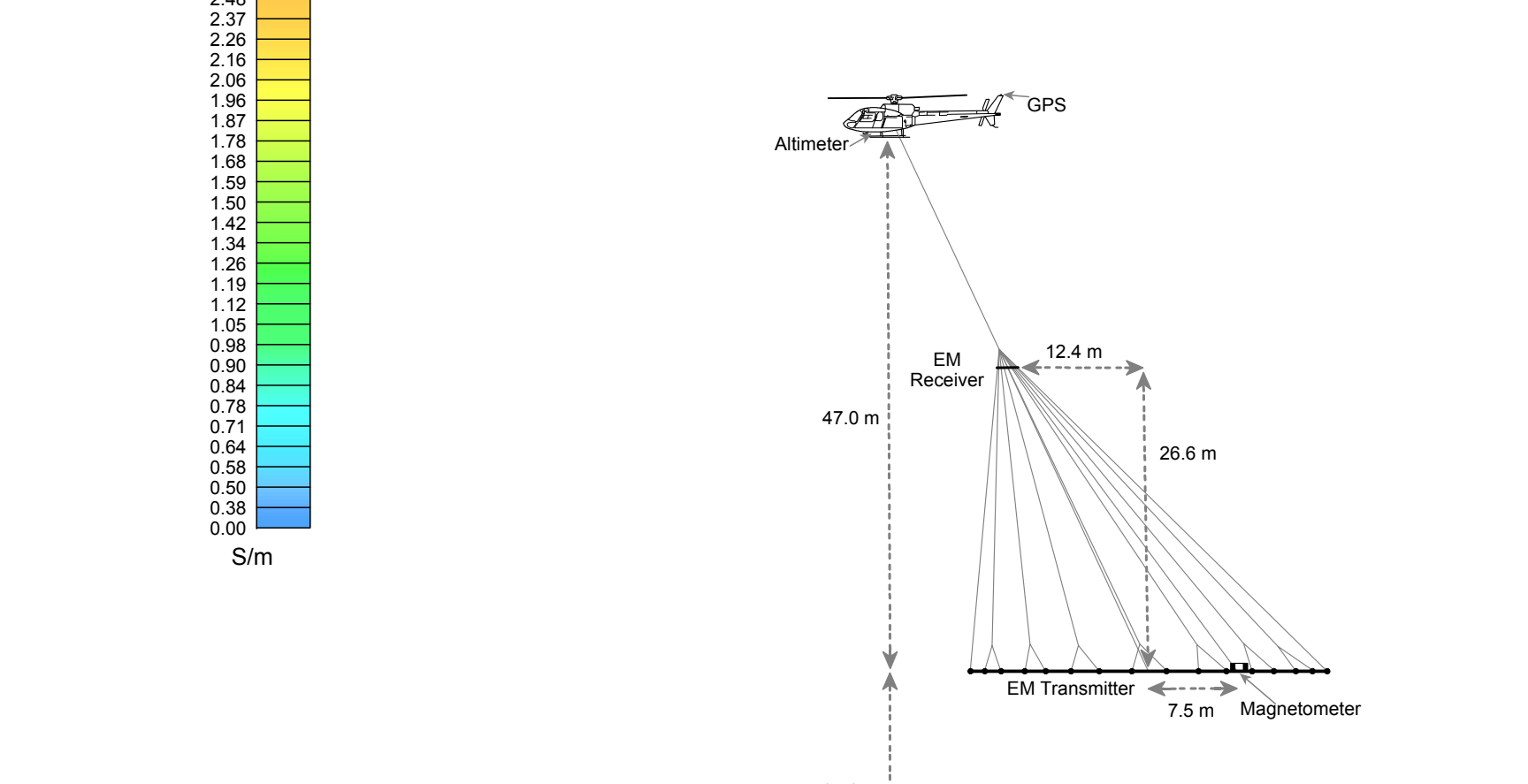
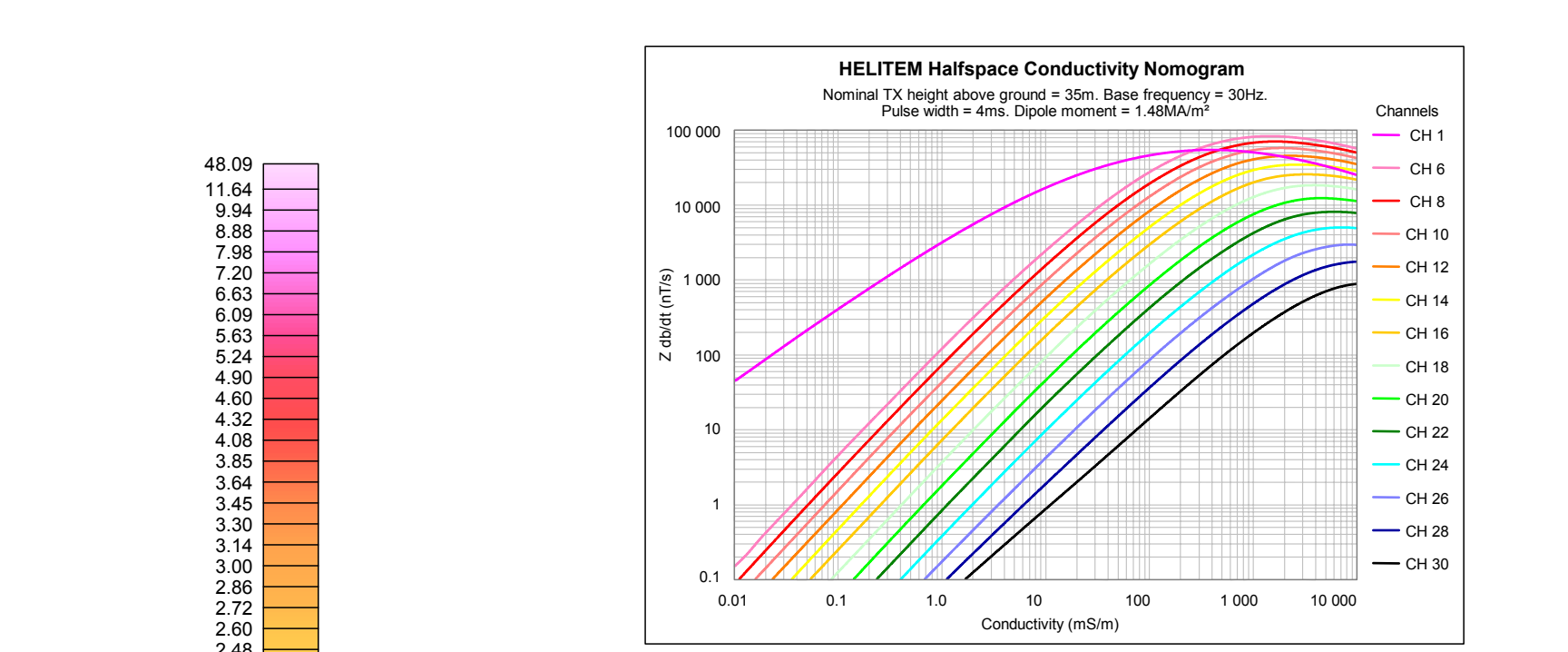
Apparent Conductivity
 The apparent conductivity values were derived from selected early, middle and late channels (6, 14 and 22) of the off-time signal, fitted to a homogeneous half-space model. This is performed using a look-up table that contains the response over a range of half-space conductivities and altimeter heights as depicted in the nomogram below.

Electromagnetic Decay Constant
 Decay constant (T_{dc}) values are obtained by fitting the data from selected early, middle and late channel ranges (6 to 6, 14 to 17 and 22 to 25) of the off-time signal to a single exponential. The decay constant indicates the relative strength of the conductor. In semi-log space, the slope of the function will reflect the exponential decay rate of the transient field and therefore the strength of the conductivity. A slow rate of decay, reflecting a high conductivity, will be represented by a high decay constant value.

Magnetic
 The magnetic field was sampled 10 times per second using a split-beam cesium vapour magnetometer (sensitivity = 0.005 nT) mounted on the transmitter loop towed behind the aircraft. The magnetic attitude of the transmitter loop control and receiver line attitude are measured and recorded. The magnetic data are recorded in a continuous stream for each of the three components. The International Geomagnetic Reference Field (IGRF) defined at a mean GPS altitude (1073.6 m) for a constant magnetic date (April 24th, 2015) was then removed. Removal of the IGRF, representing the magnetic field of Earth's core, produces a residual component essentially to magnetizations within Earth's crust.

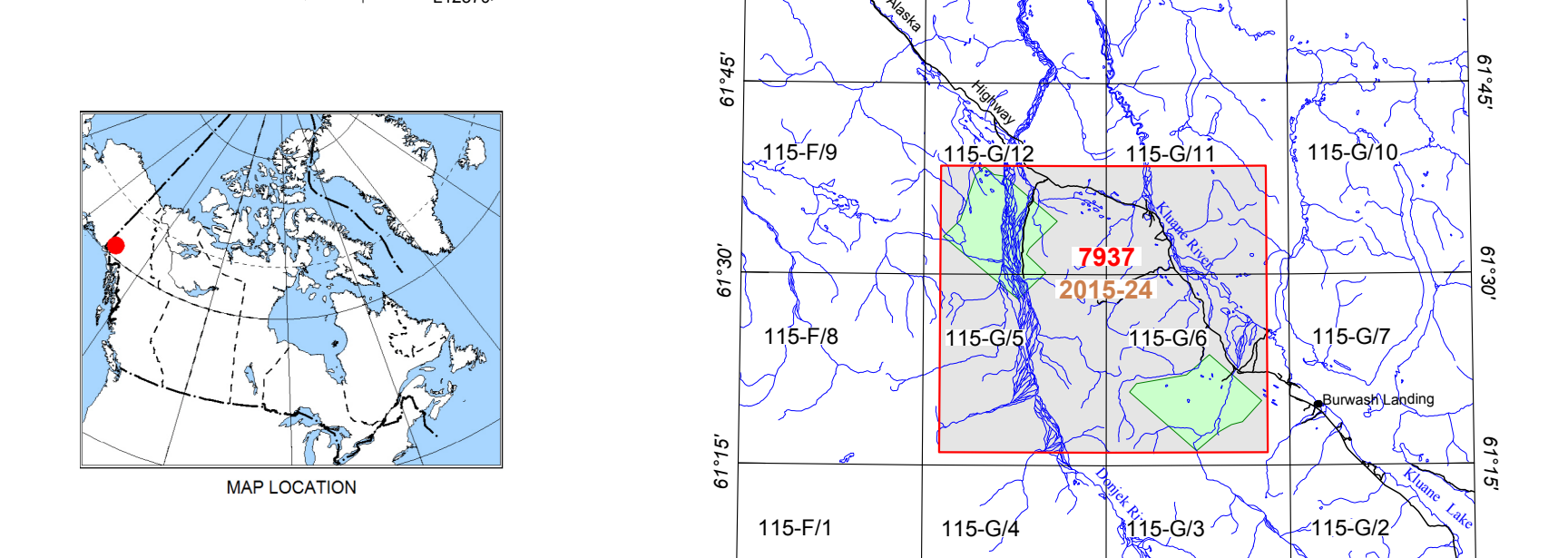
The first vertical derivative of the magnetic field is the rate of change of the magnetic field in the vertical direction. Computation of the first vertical derivative removes long-wavelength features of the magnetic field and significantly improves the resolution of closely spaced and sub-surface anomalies. A property of first vertical derivative maps is the coincidence of the zero-value contour with vertical contacts at high magnetic latitudes (Hoot, 1965). The first vertical derivative of the magnetic field was calculated by the Fast Fourier Transform on the gridded total magnetic field with a grid cell size of 62.5 m.

References
 Hoot, P.J., 1965. Gradient measurements in aeromagnetic surveying. Geophysics, v. 30, p. 891-902.



PLANIMETRIC SYMBOLS

Project Limit	—
Drainage	—
Topographic Contour	—
Building	—
Road	—
Trail	—
Mine/Quarry	—
Flight Path	—



Sheet Titles

Sheet 1: Time Decay Constant (Tau=2) - Early Channels (6 to 6)
Sheet 2: Time Decay Constant (Tau=2) - Mid Channels (14 to 17)
Sheet 3: Time Decay Constant (Tau=2) - Late Channels (22 to 25)
Sheet 4: Apparent Conductivity - Early Channel (6)
Sheet 5: Apparent Conductivity - Mid Channel (14)
Sheet 6: Apparent Conductivity - Late Channel (22)
Sheet 7: Residual Total Magnetic Field
Sheet 8: First Vertical Derivative of the Magnetic Field

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 YGS OPEN FILE 2015-24
KLUANE LAKE WEST ELECTROMAGNETIC SURVEY
 YUKON
 Parts of NTS 115-G/5, 6, 11 and 12
APPARENT CONDUCTIVITY
MID CHANNEL (14)
 Scale 1:50 000
 0 1 2 Kilometres
 NAD 83 UTM zone 7N
 North American Datum 1983
 UTM Zone 7N
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